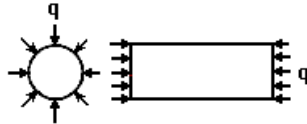


Case 20a Thin tube with closed ends under uniform external pressure, lateral and longitudinal; ends held circular

Thin tube with closed ends under uniform external pressure, lateral and longitudinal



8" NOM. (8.625" OD X 0.322" WALL X 7.981" ID) ALUMINUM 6061-T6 EXTRUDED PIPE SCHEDULE 40

Enter dimensions
and properties:

radius: $r := \frac{8.625}{2} \cdot \text{in} - \frac{.322}{2} \cdot \text{in}$ $r = 4.151 \cdot \text{in}$

length: $L_x := 34 \cdot \text{in}$

Modulus of elasticity: $E := 10 \cdot 10^6 \cdot \frac{\text{lbf}}{\text{in}^2}$

wall thickness: $t := 0.250 \cdot \text{in}$

Poisson's ratio: $\nu := 0.33$

designPressure := $750 \cdot \frac{\text{lbf}}{\text{in}^2}$

Conditions:

For the expressions in the case to be valid, the following condition must be true (i.e., condition = 1):

$\frac{r}{t} = 16.606$ condition := $\frac{r}{t} > 10$ condition = 1

Critical pressure: The general expression for q' as a function of n (the number of lobes formed by the tube in buckling) is as follows:

$$q'(n) := \frac{E \cdot \frac{t}{r}}{1 + \frac{1}{2} \cdot \left(\frac{\pi \cdot r}{n \cdot L} \right)^2} \cdot \left[\frac{1}{n^2 \cdot \left[1 + \left(\frac{n \cdot L}{\pi \cdot r} \right)^2 \right]^2} + \frac{n^2 \cdot t^2}{12 \cdot r^2 \cdot (1 - \nu^2)} \cdot \left[1 + \left(\frac{\pi \cdot r}{n \cdot L} \right)^2 \right]^2 \right]$$

To determine q' , it is necessary to determine which value of n greater than 2 provides the smallest value of q' for the given t/r . To do this, the following is defined:

$$nn := 10 \quad i := 0..nn - 1 \quad ni := i + 2$$

Find the value of n that corresponds to the smallest computed value of q' :

$$qq'i := q'(ni)$$

$$index_i := \text{if}(qq'i = \min(qq'), i, 0) \quad \underline{m} := \max(index) \quad m = 0$$

Compute q' using the correct value of n :

$$\underline{m} := 2$$

$$q'(m) = 1.048 \times 10^3 \cdot \text{psi}$$

$$SWLoadFactor := \frac{q'(m)}{\text{designPressure}} = 1.398$$

$$qq'i = \begin{pmatrix} 1.048 \times 10^3 \\ 1.9 \times 10^3 \\ 3.316 \times 10^3 \\ 5.151 \times 10^3 \\ 7.397 \times 10^3 \\ 1.005 \times 10^4 \\ 1.312 \times 10^4 \\ 1.659 \times 10^4 \\ 2.047 \times 10^4 \\ 2.476 \times 10^4 \end{pmatrix} \cdot \frac{\text{lbf}}{\text{in}^2}$$

Values of experimentally determined critical pressure range 20% above and below the theoretical value given by the expression above. A recommended probable minimum critical pressure is

$$\underline{q}' := q'(m) \cdot 0.8 \quad q' = 838.666 \cdot \text{psi}$$

References:

Ref. 19. Saunders, H. E., and D. F. Windenberg: Strength of Thin Cylindrical Shells Under External Pressure, *Trans. ASME*, vol. 53, no. 15, p. 207, 1931.

Ref. 20. von Mises, R.: Der kritische Aussendruck zylindrischer Rohre, *Z. Ver Dtsch. Ing.*, vol. 58, p. 750, 1914.

Ref. 81. Weingarten, V. I., and P. Seide: Elastic Stability of Thin-walled Cylindrical and Conical Shells Under Combined External Pressure and Axial Compression, *AIAA J.*, vol. 3, no. 5, May 1965.

Ref. 109. Johnson, B. G. (ed.): "Guide to Stability Design Criteria for Metal Structures," 3rd ed., Structural Stability Research Council, John Wiley & Sons, Inc., 1976.



End of Section

Von Mises Formula (Ross; 2nd edition page 101)

$$a := r \quad l := L$$

$$PVonMises_{cr}(n) := \frac{E \cdot \frac{t}{a}}{n^2 - 1 + 0.5 \cdot \left(\frac{\pi \cdot a}{l}\right)^2} \cdot \left[\frac{1}{\left[n^2 \cdot \left(\frac{l}{\pi \cdot a}\right)^2 + 1 \right]^2} + \frac{t^2}{12 \cdot a^2 \cdot (1 - \nu^2)} \cdot \left[n^2 - 1 + \left(\frac{\pi \cdot a}{l}\right)^2 \right] \right]$$

$$PPVonMises_{crj} := PVonMises_{cr}(n_j)$$

$$PPVonMises_{cr0} = 904.752 \frac{\text{lbf}}{\text{in}^2}$$

$$SWLoadFactor := \frac{PPVonMises_{cr0}}{\text{designPressure}} = 1.206$$

$$0.8 PVonMises_{cr}(m) = 723.802 \cdot \frac{\text{lbf}}{\text{in}^2}$$

$$PPVonMises_{crj} = \begin{pmatrix} 904.752 \\ 1.698 \times 10^3 \\ 3.112 \times 10^3 \\ 4.947 \times 10^3 \\ 7.193 \times 10^3 \\ 9.848 \times 10^3 \\ 1.291 \times 10^4 \\ 1.638 \times 10^4 \\ 2.026 \times 10^4 \\ 2.455 \times 10^4 \end{pmatrix} \cdot \frac{\text{lbf}}{\text{in}^2}$$

Windenburg and Trilling Formula aka David Taylor Model Basin formula (Ross; 2nd edition page 102)

$$PW_{andTcr} := \frac{2.42 \cdot E \cdot \left(\frac{t}{2 \cdot a}\right)^{\frac{5}{2}}}{(1 - \nu^2)^{0.75} \cdot \left[\left(\frac{1}{2 \cdot a}\right) - 0.447 \cdot \left(\frac{t}{2 \cdot a}\right)^{\frac{1}{2}}\right]}$$

$$PW_{andTcr} = 1.033 \times 10^3 \cdot \frac{\text{lbf}}{\text{in}^2}$$

$$.0549 \cdot \text{MPa} = 7.963 \cdot \frac{\text{lbf}}{\text{in}^2}$$

$$0.8 \cdot PW_{andTcr} = 826.579 \cdot \frac{\text{lbf}}{\text{in}^2}$$

Windenburg and Trilling Thinness Ratio (Ross; 2nd edition page 105)

$$d := 2 \cdot a \quad \sigma_{yp} := 40000 \cdot \frac{\text{lbf}}{\text{in}^2}$$

$$\lambda := \left[\frac{\left(\frac{1}{d}\right)^2}{\left(\frac{t}{d}\right)^3} \right]^{0.25} \cdot \left(\frac{\sigma_{yp}}{E} \right)^{.5} \quad \lambda = 1.771 \quad \frac{1}{\lambda} = 0.565$$